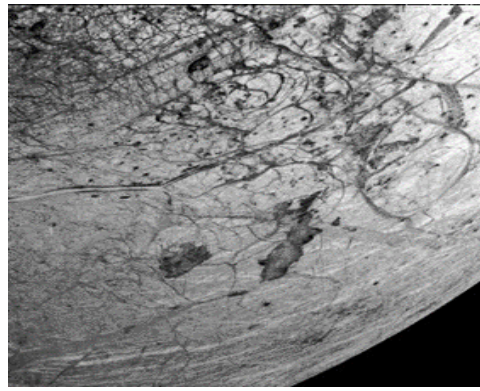
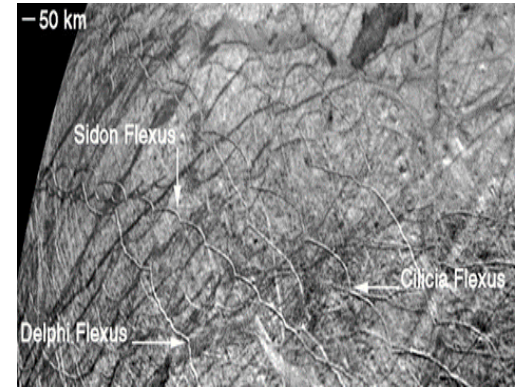


Thera Macula

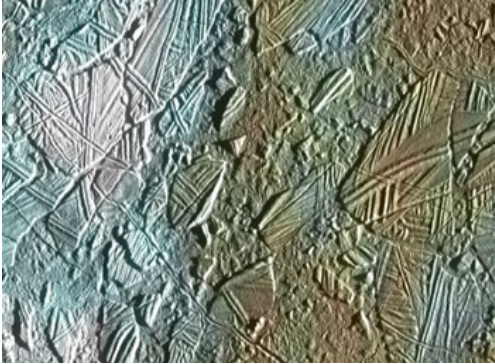


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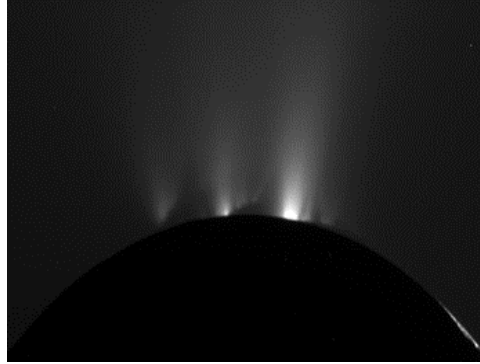


Cycloids

Europa



Conamara Chaos <Europa>



Jets



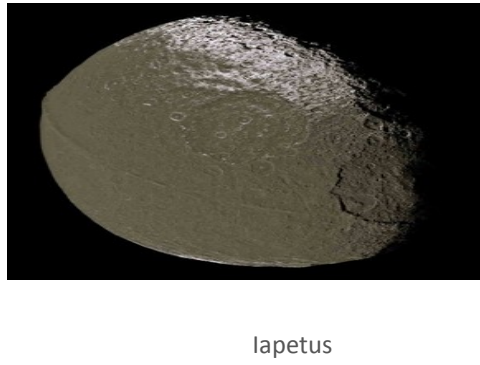
Plumes

Enceladus



Tiger Stripes

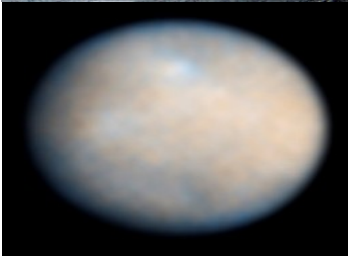
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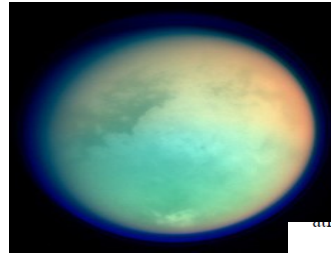
Iapetus



Triton



Ceres



Titan

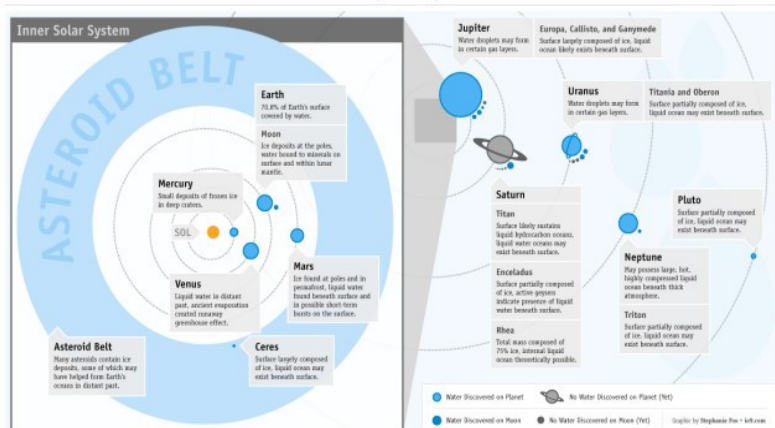


atmospheric pressure. It can also deposit

Table 1: Phases and characteristics of Water Ice [6]

Phase	Characteristics
Amorphous ice	Amorphous ice as an ice which does not have crystal structure. This ice occurs in three forms; low density (LDA) formed at atmospheric pressure, or below, high density (HDA) and very high density amorphous ice (VHDA), forming at higher pressures. LDA forms by extremely quick cooling of liquid water ("hyperquenched glassy water", HGW), by depositing water vapour on very cold substrates ("amorphous solid water", ASW) or by heating high density forms of ice at ambient pressure.
Ice I _h	Normal hexagonal crystalline ice. Almost all ice in the Earth's biosphere is ice I _h with small amount of I _h .
Ice I _c	Metastable cubic crystalline variant of ice. The oxygen atoms are arranged in a diamond structure, made at 130-150 K, and is stable upto 200K, when it transform to ice I _h . It is sometimes present in the upper atmosphere.
Ice 2	A rhombohedral crystalline form with highly ordered structure. Formed from ice I _h by compressing it at temperature of 190-210 K. When heated it undergoes transformation to ice 3.
Ice 3	A tetragonal crystalline ice, formed by cooling water down to 250 K a 300 MPa, least dense of the high pressure phases and denser than the water.
Ice 4	Metastable rhombohedral phase. Does not easily form without a nucleating agent.
Ice 5	A monoclinic crystalline phase, formed by cooling water to 253 K at 500 MPa. Most complicated structure of all.
Ice 6	A tetragonal crystalline phase, formed by cooling water to 270 K at 1.1 GPa, shows Debye relaxation.
Ice 7	A cubic phase. The hydrogen atom's position is disordered, the material shows Debye relaxation. The hydrogen bonds form two interpenetrating lattices.
Ice 8	A more ordered version of Ice 7, where the hydrogen atoms assume fixed positions, formed from ice 7 by cooling it beyond 5°C.
Ice 9	A tetragonal metastable phase, formed gradually from ice 3 by cooling it from 208 K to 165 K, stable below 140 K and pressures between 200 and 400 MPa. It has density of 1.16 g/cm ³ , slightly higher than ordinary ice.
Ice 10	Proton ordered symmetric ice, forms at about 70 GPa.
Ice 11	An orthorhombic low temperature equilibrium form of hexagonal ice, which is ferroelectric.
Ice 12	A tetragonal metastable dense crystalline phase, it is observed in the phase space of ice V and ice VI. It can be prepared by heating high-density amorphous ice from 77 K to about 183 K at 810 MPa.
Ice 13	A monoclinic crystalline phase, formed by cooling water below 130 K at 500 MPa. The proton-ordered from of ice 5.
Ice 14	An orthorhombic crystalline phase, formed below 118 K at 1.2 GPa. The proton-ordered from of ice 12.
Ice 15	The predicted but no proven proton ordered form of ice 6, thought to be formed by cooling water to around 108-80 K at 1.1 Gpa

Our Watery Solar System



Name of Satellite/Rover	Date commissioned	Instruments/What they do	Where it's been
Mars Reconnaissance Orbiter	10 March 2006, 21:24:00 UTC	HiRISE - The High Resolution Imaging Science Experiment camera is a 0.3 m reflecting telescope, the largest ever carried on a deep space mission, and has a resolution of 1 micrometer (μrad), or 0.3 m from an altitude of 300 km. Takes truly near pictures. CTX - The Context Camera (CTX) provides gray-scale images (500 to 800 nm) with a resolution up to about 6 m. CTX is designed to provide context maps for the targeted observations of HiRISE and CRISM, and is also used to mosaic large areas of Mars, monitor a number of locations for changes over time, and to acquire stereo (3D) coverage of key regions and potential future landing sites. MARCI : The Mars Color Imager (MARCI) is a wide-angle, relatively low-resolution camera that views the surface of Mars in five visible and two ultraviolet bands. Each day, MARCI collects about 84 images and produces a global map with pixel resolutions of 1 to 10 km. CRISM : The Compact Reconnaissance Imaging Spectrometer for Mars (CRISM) instrument is a visible and near infrared (VNIR) spectrometer that is used to produce detailed maps of the surface mineralogy of Mars. MCS : The Mars Climate Swander (MCS) is a spectrometer with one visible/near infrared channel (0.3 to 3 μm) and eight far infrared (11 to 50 μm) channels. These channels were selected to measure temperature, pressure, water vapor and dust levels. MSS observes the atmosphere on the horizon of Mars (as viewed from MRO) by breaking it up into vertical slices of tall, thin measurements with each slice at 1 km (3 mi) increments. SHARAD : MRO's Shallow Subsurface Radar (SHARAD) experiment is designed to probe the internal structure of the Martian polar ice caps. It also gathers planet-wide information about underground layers of ice, rock and possibly liquid water that might be accessible from the surface. SHARAD uses HF radio waves between 15 and 25 MHz.	Mars
Mars Global Surveyor	7 November 1996, 17:00 UTC // November 2, 2006	Five scientific instruments fly onboard Mars Global Surveyor [4] MOC – the Mars Orbiter Camera, operated by Malin Space Science Systems[5] MOLA – the Mars Orbiter Laser Altimeter TES - the Thermal Emission Spectrometer MAG/ER – a Magnetometer and electron reflectometer ISORS UltraSable Oscillator For Doppler measurements	Mars
Opportunity	July 7, 2003	Panoramic Camera (PanCam) - examines the texture, color, mineralogy, and structure of the local terrain. Navigation Camera (Navcam) - monochrome with a higher field of view but lower resolution, for navigation and driving. Miniature Thermal Emission Spectrometer (Mini-TES) - identifies promising rocks and soils for closer examination, and determines the processes that formed them. Hazcams , two B&W camera with 120 degree field of view, that provide additional data about the rover's surroundings. Misshower spectrometer (MB) MINIMOS - used for close-up investigations of the mineralogy of non-bearing rocks and soils. Alpha particle X-ray spectrometer (APXS) - close-up analysis of the abundance of elements that make up rocks. Magnets - for collecting magnetic dust particles. Microscopic Imager (MI) - obtains close-up, high-resolution images of rocks and soils. Rock Abrasion Tool (RAT) - exposes fresh material for examination by instruments on board.	Mars
Spirit	June 10, 2003—22 March 2010	Everything on Opportunity	Mars
Curiosity	26 November 2011	Mart Camera (MarCam)-The MarCam system provides multiple spectra and true-color imaging with two cameras.[56] The cameras can take true-color images at 1600–1200pixels and up to 10 frames per second hardware-compressed, video at 720p. ChemCam -ChemCam is a suite of remote sensing instruments, and as the name implies, ChemCam is actually two different instruments combined as one: a laser-induced breakdown spectroscopy (LIBS) and a Remote Micro Imager (RMI) telescope. NavCam, HaCaCam, REMS - REMS comprises instruments to measure the Mars environment: humidity, pressure, temperature, wind speeds, and ultraviolet radiation.[76] It is a meteorological package that includes an ultraviolet sensor provided by the Spanish Ministry of Education and Science. MARIE - MARIE is a camera on the rover's robotic arm, and acquires microscopic images of rock and soil. MARIE 1 can take true-color images at 1600–1200 pixels with a resolution as high as 14.5micrometers per pixel. APXS ChemMin - ChemMin is the Chemistry and Mineralogy X-ray powder diffraction and fluorescence instrument [86] ChemMin is one of four spectrometers. It can identify and quantify the abundance of the minerals on Mars. SAM -The SAM instrument suite analyzes organics and gases from both atmospheric and solid. DRT - DUST REMOVAL TOOL. RAD - This instrument was the first of ten MSU instruments to be launched in. Its first task was to characterize the broad spectrum of radiation environment found inside the spacecraft during the cruise phase. DAN - A pulsed sodium-laser instrument [95][96] and detector for measuring hydrogen or ice and water at or near the Martian surface. MARDI -During the descent to the Martian surface, MARDI took color images at 1600–1200 pixels with a 1.3-millisecond exposure time starting at distances of about 3.7 km (2.3 mi) to near 5 m (16 ft) from the ground, at a rate of four frames per second for about two minutes.	Mars
Phoenix	May 25, 2008— November 2, 2008 (lost communication), November 10, 2008	The Surface Stereo Imager (SSI) was the primary camera on the spacecraft. The Thermal and Evolved Gas Analyzer (TEGA) is a combination of a high-temperature furnace with a mass spectrometer. It was used to bake samples of Martian dust and determine its content.	Mars
Juice (Jupiter Icy Moons Explorer)	2022	JANUS MAIS UVS SWI GALA RIME IMAG PEP RPW1 3GM PRIDE	Jupiter
Mars Express	2 June 2003,	Visible and Infrared Mineralogical Mapping Spectrometer (OMEGA/Observatoire pour le Minéralogie, l'Eau, les Glaces et l'Activité) - France - Determines mineral composition of the surface up to 100 m resolution. It mounted inside pointing out the top face [14] Instrument mass: 26.6 kg [15] Ultraviolet and Infrared Atmospheric Spectrometer (SPICAM) - France - Assesses elemental composition of the atmosphere. It mounted inside pointing out the top face. Instrument mass: 4.7 kg [15] Sub-Surface Sounding Radar Altimeter (MARSIS) - Italy - A radar altimeter used to assess composition of sub-surface around at search for frozen water. It mounted in the body and also scans the surface, and also incorporates the two 20 m antennas. Instrument mass: 13.7 kg [15] Planetary Fourier Spectrometer (PFS) - Italy - Makes observations of atmospheric temperature and pressure (observations suspended in September 2005). It mounted inside pointing out the top face. [16] currently working. Instrument mass: 30.8 kg [15] Analyzer of Space Plasmas and Energetic Atoms (ASPERA) - Sweden - Investigates interactions between upper atmosphere and solar wind. It mounted on the top face. Instrument mass: 7.9 kg [15] High Resolution Stereo Camera (HRSC) - Germany - Produces color images with a 2 m resolution. It mounted inside the spacecraft body, aimed through the top face of the spacecraft, and is used during Mars operations. Instrument mass: 20.4 kg [15] Mars Express Lander Communications (MELACOM) - UK - Allows Mars Express to act as a communication relay for landers on the Martian surface. (Has been tested with Mars Exploration Rover, and was used to support the landing of NASA's Phoenix mission) Mars Radio Science Experiment (MARS) - Uses radio signals to investigate atmosphere, surface, subsurface, gravity and solar corona density during solar conjunctions. It uses the communications subsystem itself.	Mars

Galileo October 18, 1989, Se ptember 21, 2003

Solid State Imager (SSI)[edit] = The SSI was an 800-by-800-pixel solid state camera consisting of an array of silicon sensors called a "charge coupled device" (CCD). Galileo was one of the first spacecraft to be equipped with a CCD camera. (later needed). The optical portion of the camera was built as acoaxial telescope. Light was collected by the primary mirror and directed to a smaller secondary mirror that channeled it through a hole in the center of the primary mirror and onto the CCD. The CCD sensor was shielded from radiation, a particular problem with the harsh Jovian magnetosphere. The shielding was accomplished by means of a 10 mm thick layer of tantalum surrounding the CCD except where the light enters the system. An eight-position filter wheel was used to obtain images at specific wavelengths. The images were then combined electronically on Earth to produce color images. The spectral response of the SSI ranged from about 0.4 to 1.1 micrometers. The SSI weighed 7 kilograms and consumed an average 15 watts of power [1][12]**Near-Infrared Mapping Spectrometer (NIMS)**[edit]The NIMS instrument was sensitive to 0.7 to 2.2-micrometer wavelengths IR light overlapping the wavelength range of the SSI. The telescope associated with NIMS was all reflective (using only mirrors and no lenses) with an aperture of 229 mm. The spectrometer of NIMS used a grating to disperse the light collected by the telescope. The dispersed spectrum of light was focused on detectors of radium antimonide and silicon. The NIMS weighed 18 kilograms and used 12 watts of power on average [1][13]**Ultraviolet Spectrometer / Extreme Ultraviolet Spectrometer (UVS/EUV)**[edit] The Cassini-Huygens telescope of the UVS had a 250 mm aperture and collected light from the observation target. Both the UVS and EUV instruments used a solid grating to disperse the light. This light then passed through an exit slit into photomultiplier tubes that produced pulses or "sprays" of electrons. These electron pulses were counted, and those count numbers constituted the data that were sent to Earth. The UVS was mounted on Galileo's scan platform and could be pointed to an object in inertial space. The EUV was mounted on the spin section. At Galileo's orbital, EUV observed a narrow ribbon of space perpendicular to the spin axis. The two instruments combined weighed about 9.7 kilograms and used 5.9 watts of power [15]**Photopolarimeter-Radiometer (PPR)**[edit]The PPR had seven radiometers bands. One of these used no filters and observed all incoming radiation, both solar and thermal. Another band allowed only solar radiation through. The difference between the solar-plus-thermal and the solar-only channels gave the total thermal radiation emitted. The PPR also measured in five broadband channels that spanned the spectral range from 17 to 110 micrometers. The radiometer provided data on the temperatures of Jupiter's atmosphere and satellites. The design of the instrument was based on that of an instrument flown on the Pioneer Venus spacecraft. A 100 mm aperture reflecting telescope collected light and directed it to a series of filters, and, from there, measurements were performed by the detectors of the PPR. The PPR weighed 5.0 kilograms and consumed about 5 watts of power [17][18]**Spin section/Inertial Dust Detector Subsystem (IDS)**[edit] The Dust Detector Subsystem (DDS) was used to measure the mass, electric charge, and velocity of incoming particles. The masses of dust particles that the DDS could detect go from 10–16 to 10^{−7} grams. The speed of these small particles could be measured over the range of 1 to 70 kilometers per second. The instrument could measure impact rates from 1 particle per 115 days (10 megaseconds) to 100 particles per second. Such data was used to help determine dust origin and dynamics within the magnetosphere. The DDS weighed 4.2 kilograms and used an average of 5.4 watts of power [19]**20****Energetic Particles Detector (EPD)**[edit] The Energetic Particles Detectors (EPD) was designed to measure the numbers and energies of ions and electrons whose energies exceeded about 20 keV (2.2 D). The EPD could also measure the direction of travel of such particles, and, in the case of ions, could determine their composition (whether the ion is oxygen or sulfur, for example). The EPD used silicon solid state detectors and a time-of-flight detector system to measure changes in the energetic particle population at Jupiter as a function of position and time. These measurements helped determine how the particles get their energy and how they were transported through Jupiter's magnetosphere. The EPD weighed 10.5 kilograms and used 10.1 watts of power on average [21][22]**Heavy Ion Counter (HIC)**[edit] The HIC was an effort to repackage and update versions of some parts of the flight spare of the Voyager Cosmic Ray System. The HIC detected heavy ions using stacks of angle-crystal silicon wafers. The HIC could measure heavy ions with energies as low as 6 MeV (6 p) and as high as 200 MeV (32 p) per nucleon. This range included all atomic substances between carbon and nickel. The HIC and the EUV share a communications link and, therefore, had to share observing time. The HIC weighed 9 kilograms and used an average of 2.8 watts of power [23]**24****Magnetometer (MAG)**[edit] The magnetometer (MAG) used two sets of three sensors. The three sensors allowed the three orthogonal components of the magnetic field section to be measured. One set was located at the end of the magnetometer boom and, in that position, was about 11 m from the spin axis of the spacecraft. The second set, designed to detect stronger fields, was 6 m from the spin axis. The boom was used to remove the MAG from the immediate vicinity of Galileo to minimize magnetic effects from the spacecraft. However, not all these effects could be eliminated by distancing the instrument. The rotation of the spacecraft was used to separate natural magnetic fields from engineering-induced fields. Another source of potential error in measurement came from the bending and twisting of the long-magnetometer boom. To account for these motions, a calibration coil was mounted rigidly on the spacecraft to generate a reference magnetic field during calibrations. The magnetic field at the surface of the Earth has a strength of about 50,000 nT. At Jupiter, the onboard (11 m) set of sensors could measure magnetic field strengths in the range from 332 to ±512 nT, while the inboard (6 m) set was active in the range from ±512 to ±16,384 nT. The MAG experiment weighed 7 kilograms and used 3.9 watts of power [25]**26****Plasma Subsystem (PWS)**[edit] The PWS used seven fields of view to collect charged particles for energy and mass analysis. These fields of view covered most angles from 0 to 180 degrees, fanning out from the spin axis. The rotation of the spacecraft carried each field of view through a full circle. The PLS measured particles in the energy range from 0.9 eV to 52 keV (1 aJ to 8.3 D). The PLS weighed 13.2 kilograms and used an average of 10.7 watts of power [27]**28****Plasma Wave Subsystem (PWS)**[edit]An electric dipole antenna was used to study the electric fields of plasmas, while two search coil magnetic antennas studied the magnetic fields. The electric dipole antenna was mounted at the tip of the magnetometer boom. The search coil magnetic antennas were mounted on the high-gain antenna feed. Nearly simultaneous measurements of the electric and magnetic field spectrum allowed electrostatic waves to be distinguished from electromagnetic waves. The PWS weighed 7.1 kilograms and used an average of 9.8 watts [29][30]

Cassini-Huygens October 15, 1997,

Cassini Plasma Spectrometer (CAPS)The CAPS is a direct sensing instrument that measures the energy and electrical charge of particles that the instrument encounters, (the number of electrons and protons in the particle). CAPS will measure the molecules originating from Saturn's ionosphere and also determine the configuration of Saturn's magnetic field. CAPS will also investigate plasma in these areas as well as the solar wind within Saturn's magnetosphere.[9][20] CAPS has been turned off since June 2011 because of an electrical short circuit that occurred in the instrument. The instrument was powered on in March 2012, after 78 days a second short circuit forced the instrument to be shutdown again.[21]**Cosmic Dust Analyzer (CDA)**The CDA is a direct sensing instrument that measures the size, speed, and direction of tiny dust grains near Saturn. Some of these particles are coming from Saturn, while others may come from other star systems. The CDA on the orbiter is designed to learn more about these mysterious particles, the materials in other celestial bodies and potentially about the origins of the universe.[19]**Composite Infrared Spectrometer (CIRS)**The CIRS is a remote sensing instrument that measures the infrared waves coming from objects to learn about their temperatures, thermal properties, and compositions. Throughout the Cassini-Huygens mission, the CIRS will measure infrared emissions from atmospheres, rings and surfaces in the vast Saturn system. It will map the atmosphere of Saturn in three dimensions to determine temperature and pressure profiles with altitude, gas composition, and the distribution of aerosols and clouds. It will also measure thermal characteristics and the composition of satellite surfaces and rings.[19]**Ion and Neutral Mass Spectrometer (INMS)**The INMS is a direct sensing instrument that analyzes charged particles (like protons and heavier ions) and neutral particles (like atoms) near Titan and Saturn to learn more about their atmospheres. INMS is intended also to measure the positive ion and neutral environments of Saturn's satellites and rings.[19][22]**Imaging Science Subsystem (ISS)**The ISS is a remote sensing instrument that captures most images in visible light, and also some infrared images and ultraviolet images. The ISS has taken hundreds of thousands of images of Saturn, its rings, and its moons, for return to the Earth by radio telemetry. The ISS has a wide-angle camera (WAC) that takes pictures of large areas, and a narrow-angle camera (NAC) that takes pictures of small areas in fine detail. Each of these cameras uses a sensitive charge-coupled device (CCD) as a photoelectric wave detector. Each CCD has a 1024-square array of pixels, 12 on one side. Both cameras allow for many data collection modes, including on-chip data compression. Both cameras are fitted with optical filters that rotate on a wheel—to view different bands within the electromagnetic spectrum ranging from 0.2 to 1.1 μm.[19]**24****Dual Techeque Magnetometer (MAG)**The MAG is a direct sensing instrument that measures the strength and direction of the magnetic field around Saturn. The magnetic fields are generated partly by the intensely hot molten core of Saturn's center. Measuring the magnetic field is one of the ways to probe the core, even though it is far too hot and deep to visit. MAG aims to develop a three-dimensional model of Saturn's magnetosphere, and determine the magnetic state of Titan and its atmosphere, and the icy satellites and their role in the magnetosphere of Saturn.[19]**25****Magnetospheric Imaging Instrument (MIMI)**The MIMI is both a direct and remote sensing instrument that produces images and other data about the particles trapped in Saturn's huge magnetic field, or magnetosphere. This information will be used to study the overall configuration and dynamics of Saturn's magnetosphere, and its interaction with the solar wind. MIMI includes the Ion and Neutral Mass Spectrometer (INMS) which radiates and receives light over wavelengths from 0.4 to 190 nm. The instrument also has a sensitive Neutral Atoms (ENAs)[27]**Radar** The onboard radar is a remote active and remote passive sensing instrument that will produce maps of Titan's surface. It measures the height of surface objects (like mountains and canyons) by sending radio signals that bounce off Titan's surface and timing their return. Radio waves can penetrate the thick veil of haze surrounding Titan. The radar will listen for radio waves that Saturn or its moons may be producing.[19]**Radio and Plasma Wave Science Instrument (RPWS)**The RPWS is a direct active remote sensing instrument that records and measures radio signals coming from Saturn, including the radio waves given off by the interaction of the solar wind with Saturn and Titan. RPWS is to measure the electric and magnetic wave fields in the ionospheric medium and planetary magnetospheres. It will also determine the electron density and temperature near Titan and in some regions of Saturn's magnetosphere. RPWS studies the configuration of Saturn's magnetic field and its relationship to Saturn's Kilometric Radiation (SKR), as well as monitoring and mapping Saturn's ionosphere, plasma, and lightning from Saturn (and possibly Titan) in its atmosphere.[19]**Radio Science Subsystem (RSS)**The RSS is a remote sensing instrument that uses radio antennas on Earth to observe the way radio signals from the spacecraft change as they are sent through objects, such as Titan's atmosphere or Saturn's rings, or even behind the Sun. The RSS also studies the compositions, pressures and temperatures of atmospheres and ionospheres, radial structure and particle size distribution within rings, body and system masses and gravitational waves. The instrument uses the spacecraft X-band communication link as well as S-band downlink and Ka-band uplink and downlink.[19]**3****Visible and Infrared Mapping Spectrometer (VIMS)**The VIMS is a remote sensing instrument that captures images of the ultraviolet light reflected off an object, such as the clouds of Saturn and/or its rings, to learn more about their structure and composition. Designed to measure ultraviolet light over wavelengths from 0.3 to 190 nm, the instrument is also a valuable tool to help determine the composition, distribution, aerosol particle content and temperatures of their atmospheres. Unlike other types of spectrometers, this sensitive instrument can take both spectral and spatial readings. It is particularly adept at determining the composition of gases. Spatial observations take a wide-by-narrow view, only one pixel tall and 64 pixels across. The spectral dimension is 1024 pixels per spatial pixel. Also, it can take many images that create movies of the ways in which the material is moved around by other forces.[19]**Visible and Infrared Mapping Spectrometer (VIMS)**The VIMS is a remote sensing instrument that captures images using visible and infrared light to learn more about the composition of most surfaces, the rings, and the atmospheres of Saturn and Titan. It is made up of two cameras or one: one used to measure visible light, the other infrared. VIMS measures reflected and emitted radiation from all atmospheres, rings and surfaces over wavelengths from 500 to 5100 nm, to help determine their compositions, temperatures and structures. It also observes thermal and daylight that passes through the rings to learn more about their structure. Scientists plan to use VIMS for long-term studies of cloud movement and morphology in the Saturn system, to determine Saturn's weather patterns.[19]

Jupiter

Saturn/Its moons